

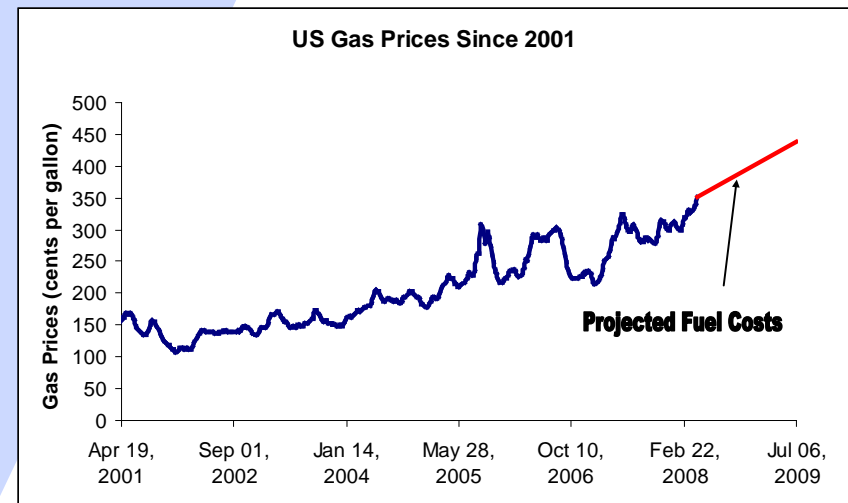
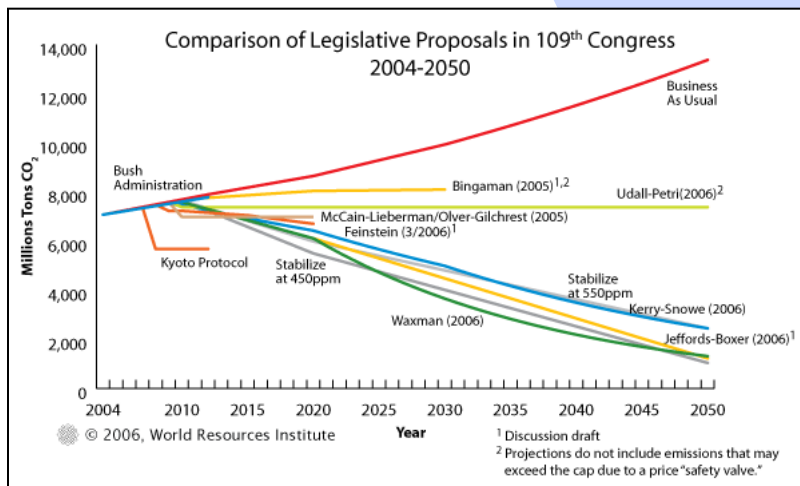
ecocopter



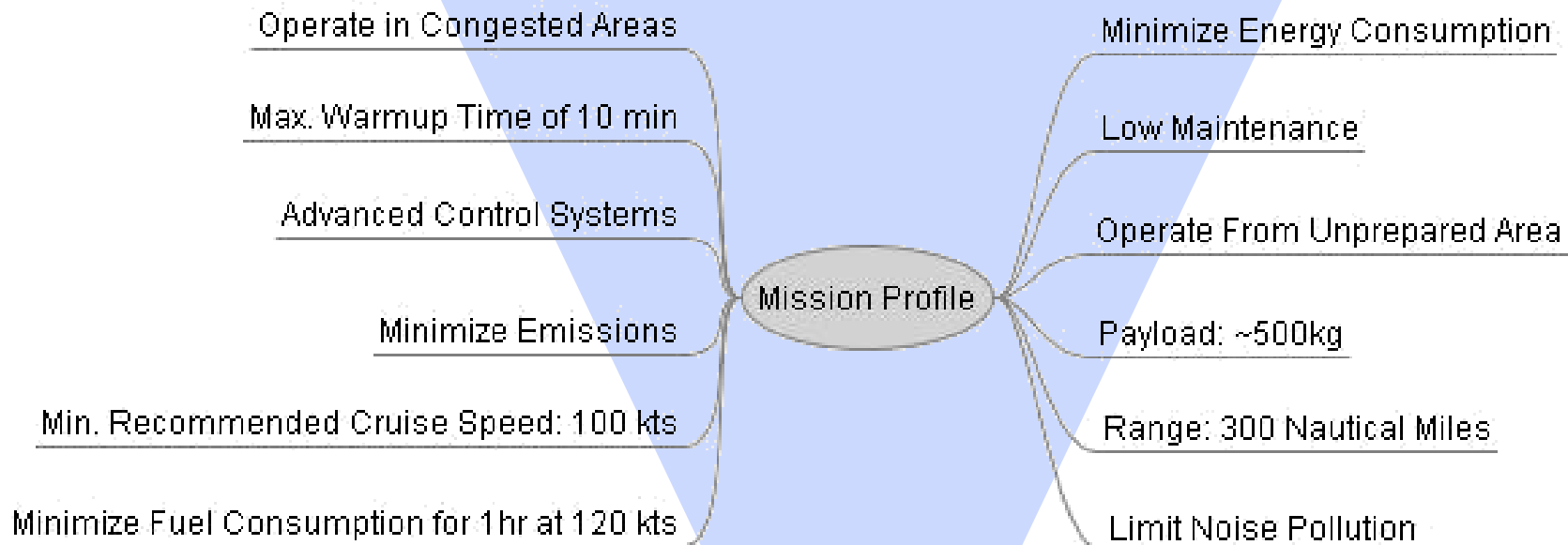
The search for a
greener tomorrow

Introduction

In today's market for vehicles there is a high demand for high efficiency, low emission vehicles and the helicopter market is no different.



Requirement Analysis



Quality Function Deployment

Quality Function Deployment	Customer Importance	HOWs (Title)																
		Figure of Merit	Gross Weight	Disk Loading	Tip Speed	Rotor Radius	Solidity	# of Blades	Taper	Pitch Ranges	Rotor Inertia	Shaft Tilt	# of Engines	# of Main Rotors	Volume	SFC	Power Available	Blade Twist
How Much		0.8	3200 lbs	4.5 psf	675 ft/s	14.5 ft	0.054	4	2:1		20 AI		1	1	54.5 cubic ft	.3 lb/hr	375 HP	-12deg linear
Organizational Difficulty		3	3	2	3	4	4	2	1	2	3	4	5	5	3	2	2	1
Weighted Importance		136.0	239.0	125.0	164.0	114.0	53.0	237.0	38.0	8.0	125.0	25.0	200.0	261.0	128.0	297.0	127.0	112.0
Relative Importance																		
Absolute Difficulty		45.3	79.7	62.5	65.6	28.5	13.3	118.5	38.0	4.0	41.7	6.3	40.0	52.2	42.7	148.5	63.5	112.0
Relative Difficulty																		

Overall Metric

- Need a way to rate different aircraft configurations by the requirements of the RFP
- Used an overall metric, combining four sub-metrics based on the results of the QFD
- Sub-metrics used target values to provide an rating for each section
- Sub-metrics
 - Efficiency Rating (EfR)
 - Mission Rating (MiR)
 - Emission Rating (EmR)
 - Cost Rating (CR)

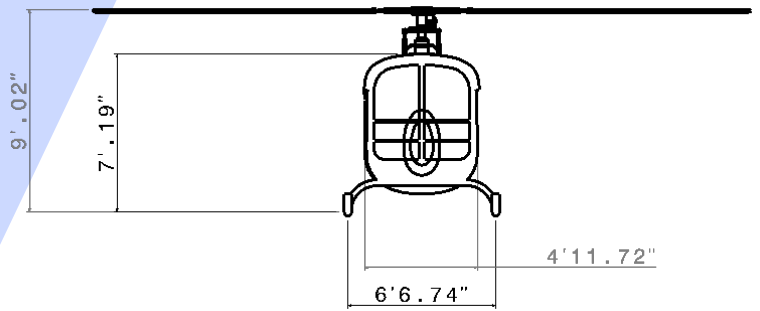
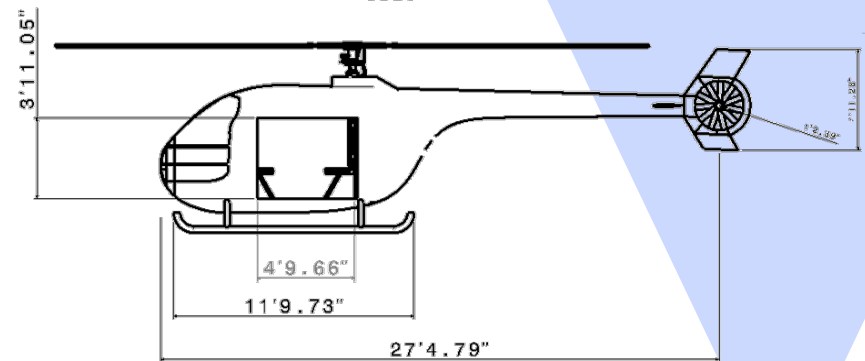
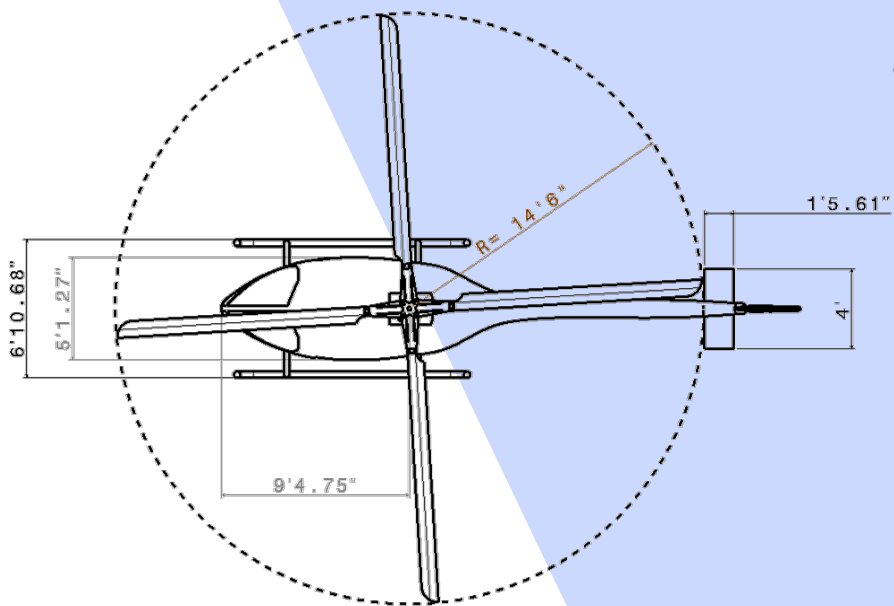
	Overall Metric
Ecocopter	0.785
MD 500E	0.700
MD 520	0.677
EC 120	0.671
Bell 206L4	0.658
EC 130	0.641
Bell 407	0.609

Vehicle Sizing and Performance

- **Baseline Vehicle Configuration**
 - Focused on high efficiency and low emissions
 - Single main rotor helicopter
 - Fenestron anti-torque system
 - Skid landing gear
- **Methodology**
 - CIRADS (R_f method)
 - Design Matrix

Fuselage Dimensions		Main Rotor	
Length	8.7 m	Radius	4.42 m
Height	2.74 m	Number of Blades	4
Width	2 m	Solidity	0.05
Weights		Tip speed	206 m/s
Gross Weight	1615 kg	Tip Taper Ratio	5:3
Empty Weight	858 kg	Shaft Tilt	0°
Fuel Weight	158 kg	Disk Loading	217 N/m ²
Payload	757 kg	Twist Rates	-8° linear
Performance			
Recommended Cruise		114 kts	
Max Airspeed		159 kts	
Endurance Airspeed		71 kts	
Max ROC		630 m/min	
Range with 20min reserve		411 Nmi	
Endurance with 20min reserve		4.5 hr	

Vehicle Sizing and Performance



Key Design Features



Rotor System

- Hingeless Rotor System
 - Use of material flexing to compensate for dissymmetry of lift, flapping, and lead/lag



- Starflex (Eurocopter AS series)
- Spheriflex (Eurocopter EC series)

Pros

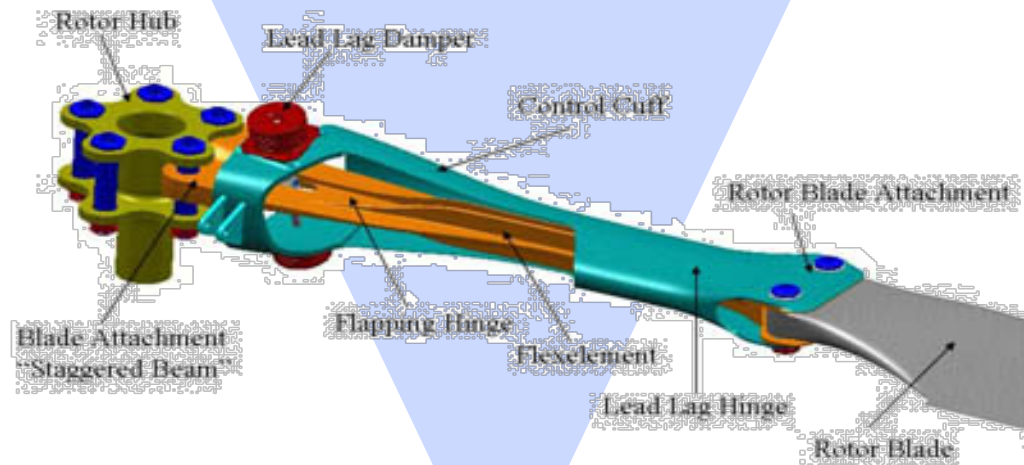
- Fewer Parts
- Simplicity
- Reduction in hub vibrations

Cons

- Cost (composite material and manufacturing)

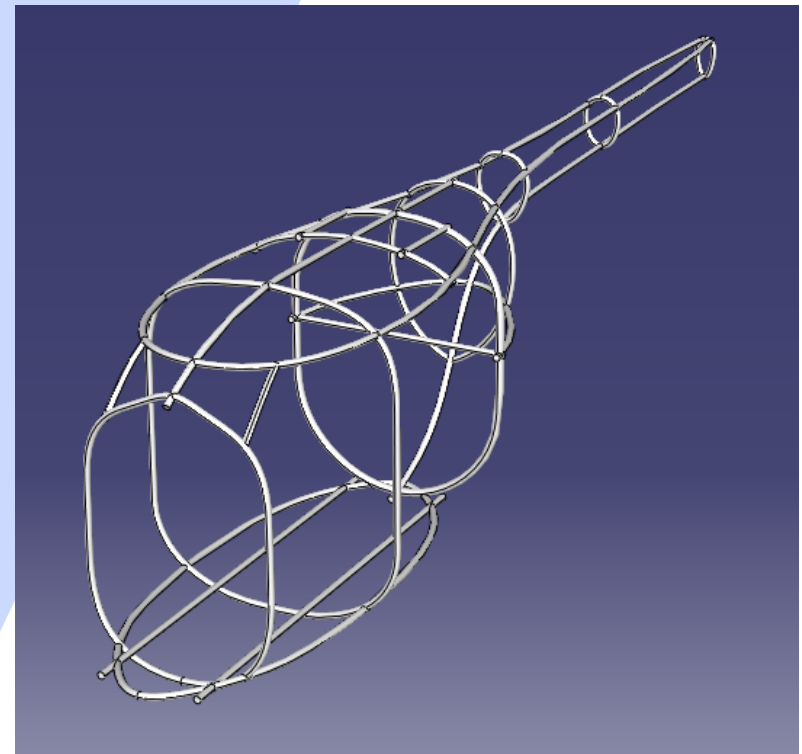
Rotor System

- Advanced Technology Rotor (ATR)
 - Hingeless and bearingless system
 - Feathering accomplished by beam twisting
 - Developed by Eurocopter
 - Modified for the EcoCopter's four blade rotor system



Structural Analysis

- Structural members to support helicopter frame
 - Maneuvering loads
 - Main rotor: +3.5g, -1.0g
 - Tail rotor
 - Crash loads
 - Frame floor impact: 4.5g
- Designed in CATIA
- Analyzed in ABAQUS



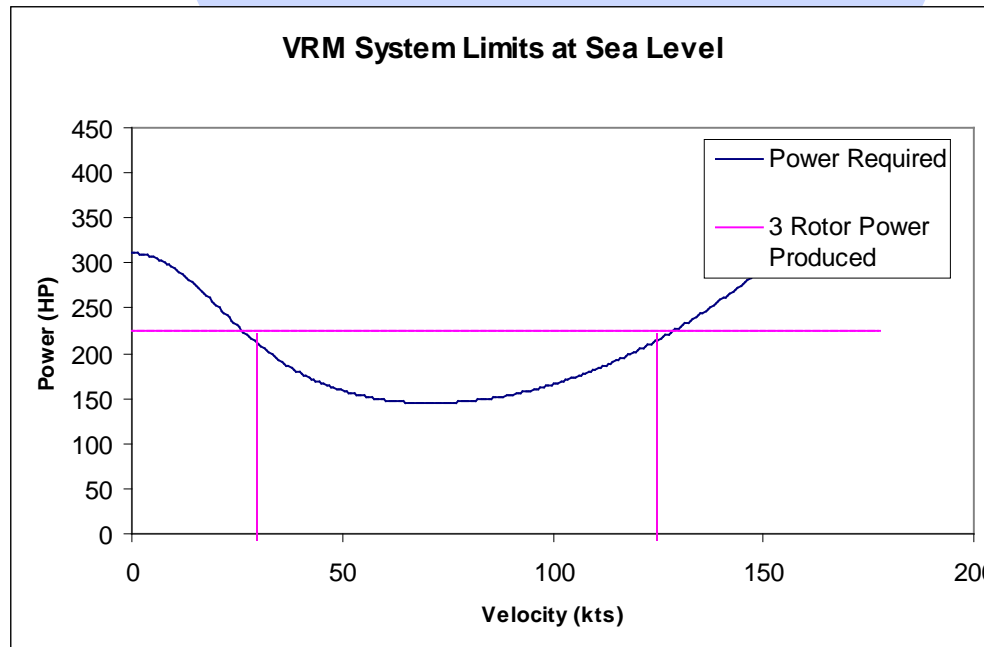
Freedom Rotapower 530

Type	Rotary
Fuels	Multi-fuel Capable
Power	300 HP
Weight	140 lb
Power Density	.47 lb/HP
SFC	.4 lb/HP-hr
Emissions	SULEV



Engine Configuration

- VRM System
 - Engine computer shuts off fuel injectors and spark plugs to inactive rotors
 - Solenoid detaches inactive rotors from camshaft
- VRM Conditions



Fuel Selection

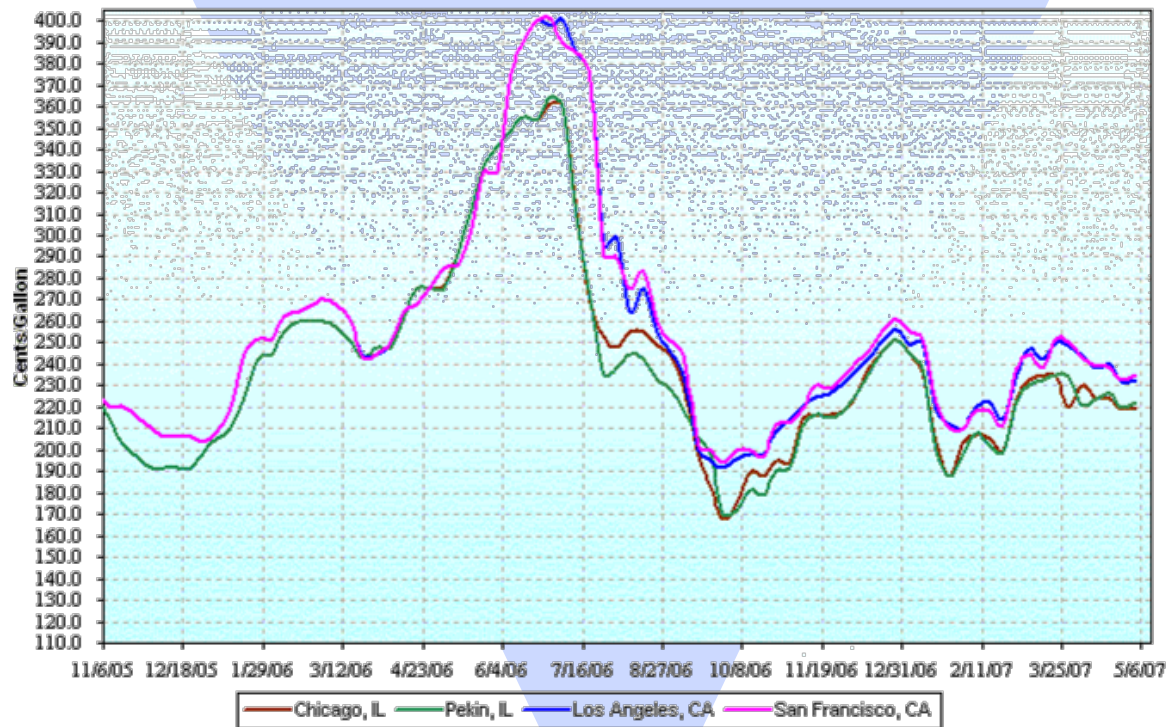
- Fuel Selected: Ethanol

Compound	Emissions (g/hp-hr)
NO _x	0.016
CO	0.03
UHC	0.0043

- 25% lower CO₂ emissions compared to Gasoline
- Renewable Resource
- Lower Cost than Gasoline

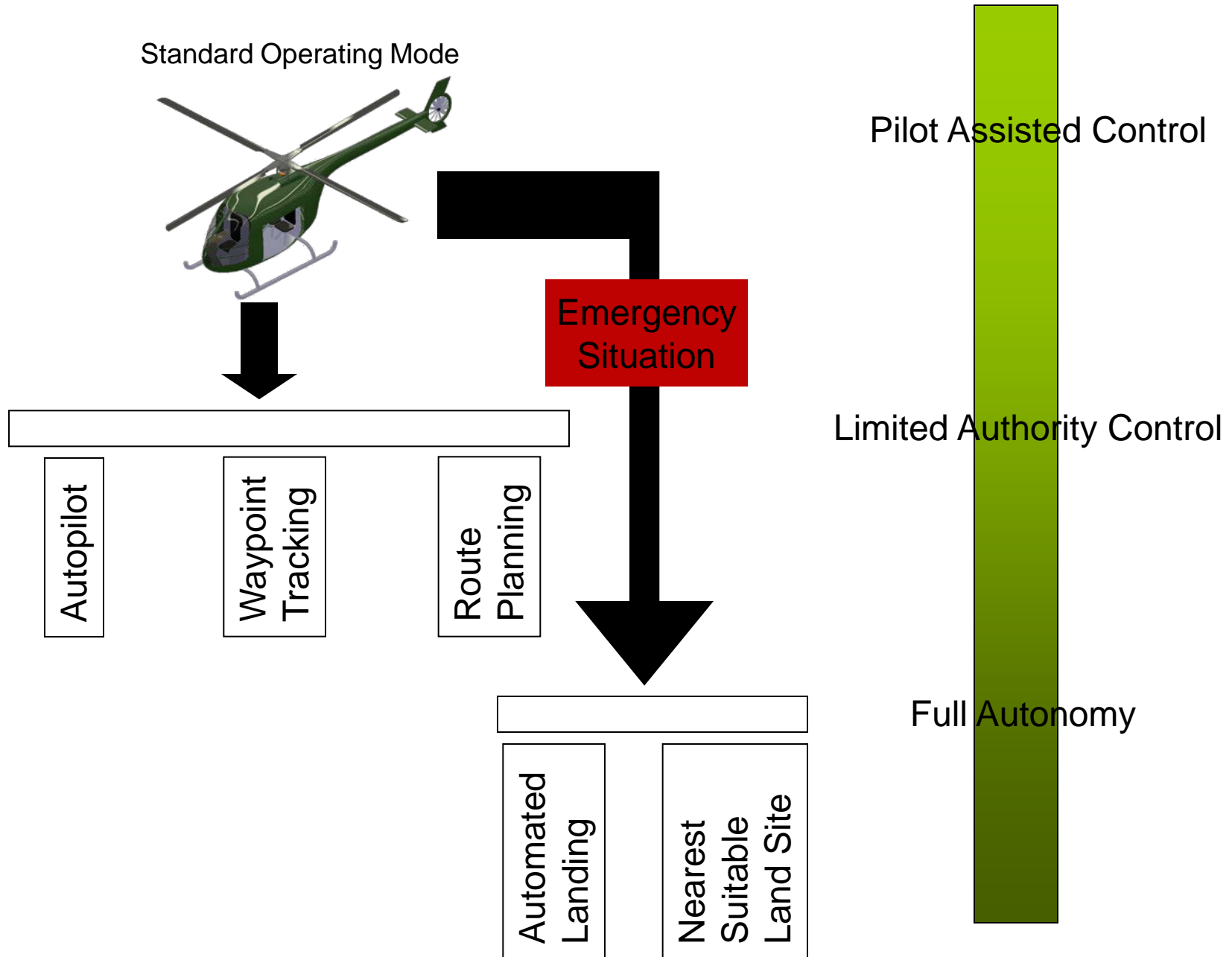
Fuel Selection

- Ethanol Costs Since 2005



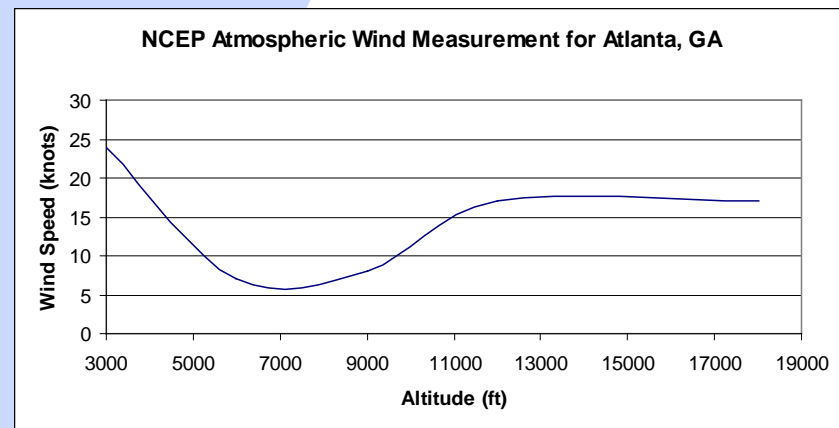
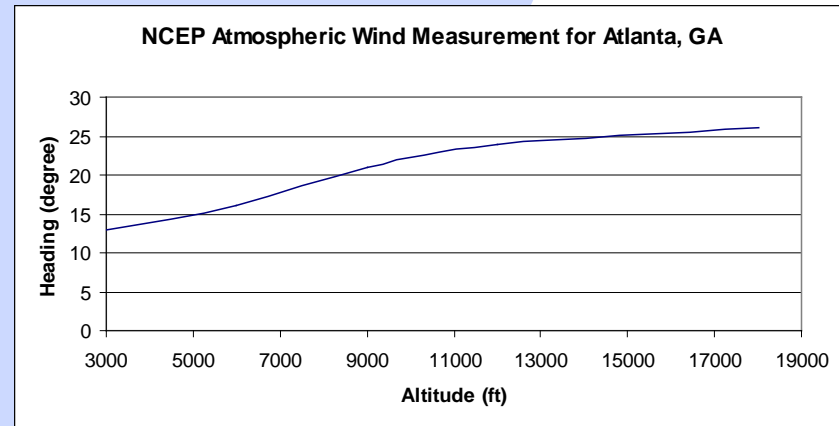
Data Source: OXY-FUEL News Price Report. 1995-2005 Hart Publications, Inc.

Control System Hierarchy



Advanced Route Planning

- NOAA NCEP Winds Aloft forecasts report wind speed variations of approximately 5-7 knots and $\sim 5^\circ$ heading variations every 3000ft



Advanced Route Planning

- Atmospheric based route planning will provide a noticeable improvement to performance

Cruise power required @ 5000ft		
Condition	Power Required	% Power Reduction
0 knot tailwind	210	0
5 knot tailwind	199.3	5.0952
10 knot tailwind	189.8	9.619

Cost Analysis

- Reviewed direct operating costs per flight hour using the Bell Cost Model using very conservative values for fuel costs

Ecocopter

DIRECT OPERATING COST		
FUEL AND LUBRICANTS		
FUEL (13.7 GAL/FH @ \$2.50/GAL)	\$34.13	
LUBRICANTS @ 3.0% OF FUEL COST	1.02	
TOTAL FUEL AND LUBRICANTS		\$35.15
AIRFRAME MAINTENANCE		
LINE MAINTENANCE (0.215 MH/FH)	\$10.75	
LIFE-LIMITED PARTS (REMOVE/REPLACE 0.018 MH/FH)	0.92	
LIFE-LIMITED PARTS COST	45.38	
OVERHAUL PARTS (REMOVE/REPLACE 0.001 MH/FH)	0.04	
OVERHAUL PARTS (COST TO OVERHAUL)	6.35	
UNSCHEDULED REMOVE/REPLACE LABOR (0.563 MH/FH)	28.14	
UNSCHEDULED PARTS COST	33.43	
TOTAL AIRFRAME MAINTENANCE		\$125.01
POWERPLANT OVERHAUL AND MAINTENANCE		\$54.70
TOTAL DIRECT OPERATING COST		\$214.87

- Published operating cost values for competitors were outdated and values based on current fuel prices were calculated

Helicopter	Operating Cost (\$2001)	Operating Cost (\$2008)
Ecocopter	N/A	\$215
EC-120	\$220	\$297
MD500	\$220	\$284

Conclusion

Everything you would expect from a “green” helicopter...

- Superior Engine Performance
 - High Efficiency
 - Low Emissions
 - Multi-fuel capable
- Improved Safety
 - Fenestron Anti-torque System
 - High Autorotative Index
 - Emergency Autonomous Control



... and more.